

**X2000**

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# **Advanced Deep Space System Development Program**

## **FIRST DELIVERY**

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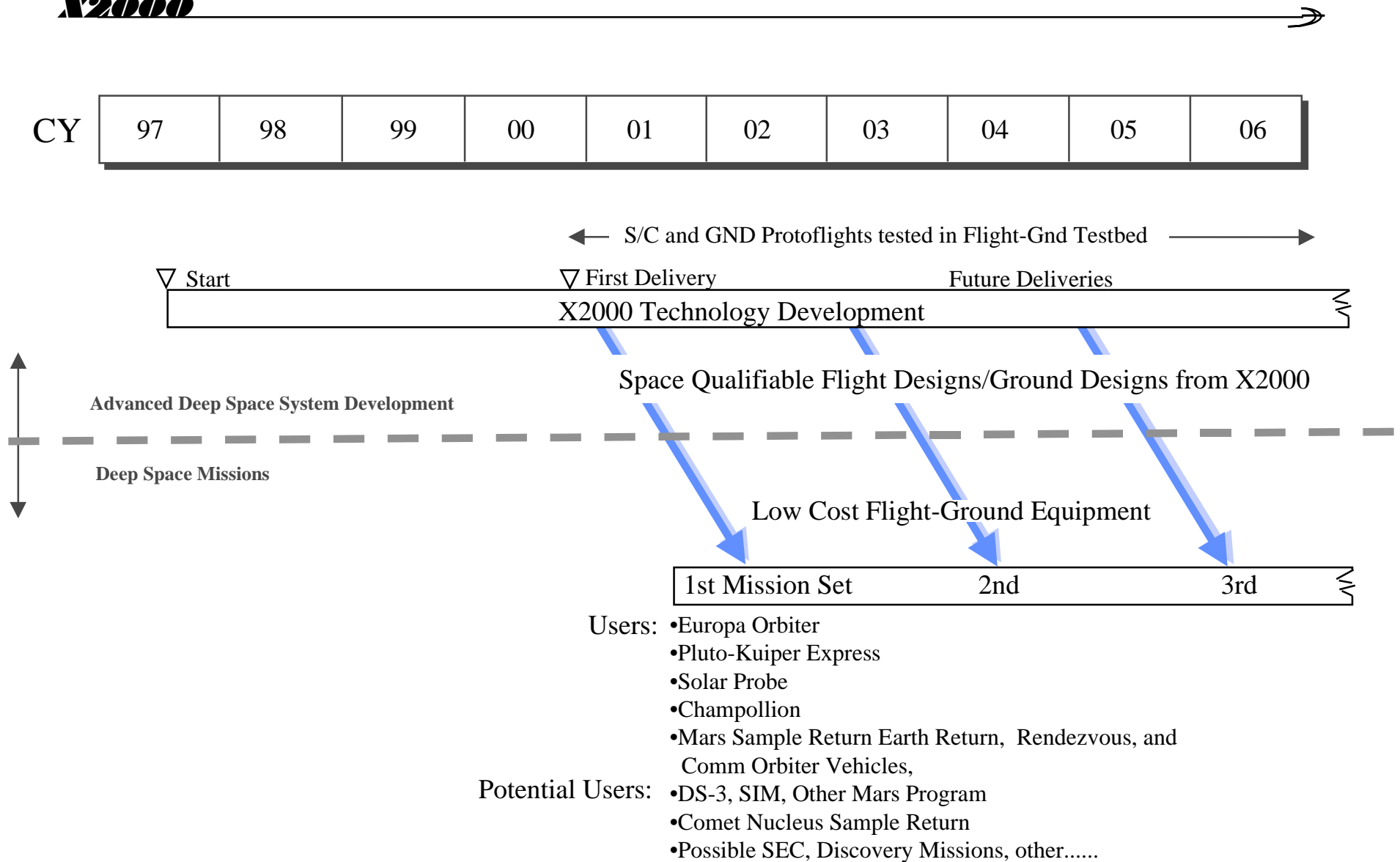
**Advanced Deep Space System Development Program Workshop on Advanced Spacecraft Technologies**

**Pasadena, CA**

**June 3, 1997**

# Flow of Advanced Deep Space System Development Deliveries

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SEC = Sun Earth Connection

GND= Ground

S/C= Spacecraft

DS-3= Deep Space 3

SIM=Space Interferometry Mission

# Advanced Deep Space System Development

## 1st Delivery

**A2000**



### **AN INTEGRATED and QUALIFIED engineering model FLIGHT and GROUND SYSTEM**

- Flight System Micro-electronics
  - Computer and memory
    - Neural Network
    - Digital Signal Processor
  - Power & Pyro Switching
  - Sensor/Instrument I/O
  - Scaleable, modular, long life
  - RAD Hardened Designs and Parts
  - Low Temperature, low power
  - X and Ka Band Comm
  - Optical Comm -- Possible
- F/S and GND S/W with W.S:
  - Operating systems
  - Generic auto NAV, 3-Axis A/C
  - Generic F/S-GND autonomy
  - Generic F/S-GND science data processing
  - Generic GND CMD/TLM processing/display
- ARPS power source
- Other: Micro-electronic components, structure, thermal, propulsion, etc. as budget allows

# Flight System Summary

## An Example for a Europa Orbiter

**A2000**



- o Science
  - Radar sounder, LIDAR, mult. spec. WAC & MAC
- o Telecom
  - Redundant Optical Com/NAC/Laser Alt receiver
  - Single DSTT with X-band SSPA & MGA
- o Data System
  - 3D Stack MCM Computers (3)
  - Stacked memory Solid State Recorder
  - Multiple buses/variable power draw
- o Power
  - 150 W. ARPS
  - Power & pyro switching microelectronics
- o Attitude Control
  - 3 axis stabilized
  - Adv. Stellar Compass w/CPU
  - Solid state IMU
  - Sun sensors
- o Propulsion
  - 400 N dual mode w/liquid regulators
  - 20 N TVC biprop thrusters
  - Monoprop 1 N and 5 mN RCS thrusters
- o Temperature Control
  - ARPS waste heat
  - MLI blankets and louvers
  - RHU's
  - Electric heaters

### Performance

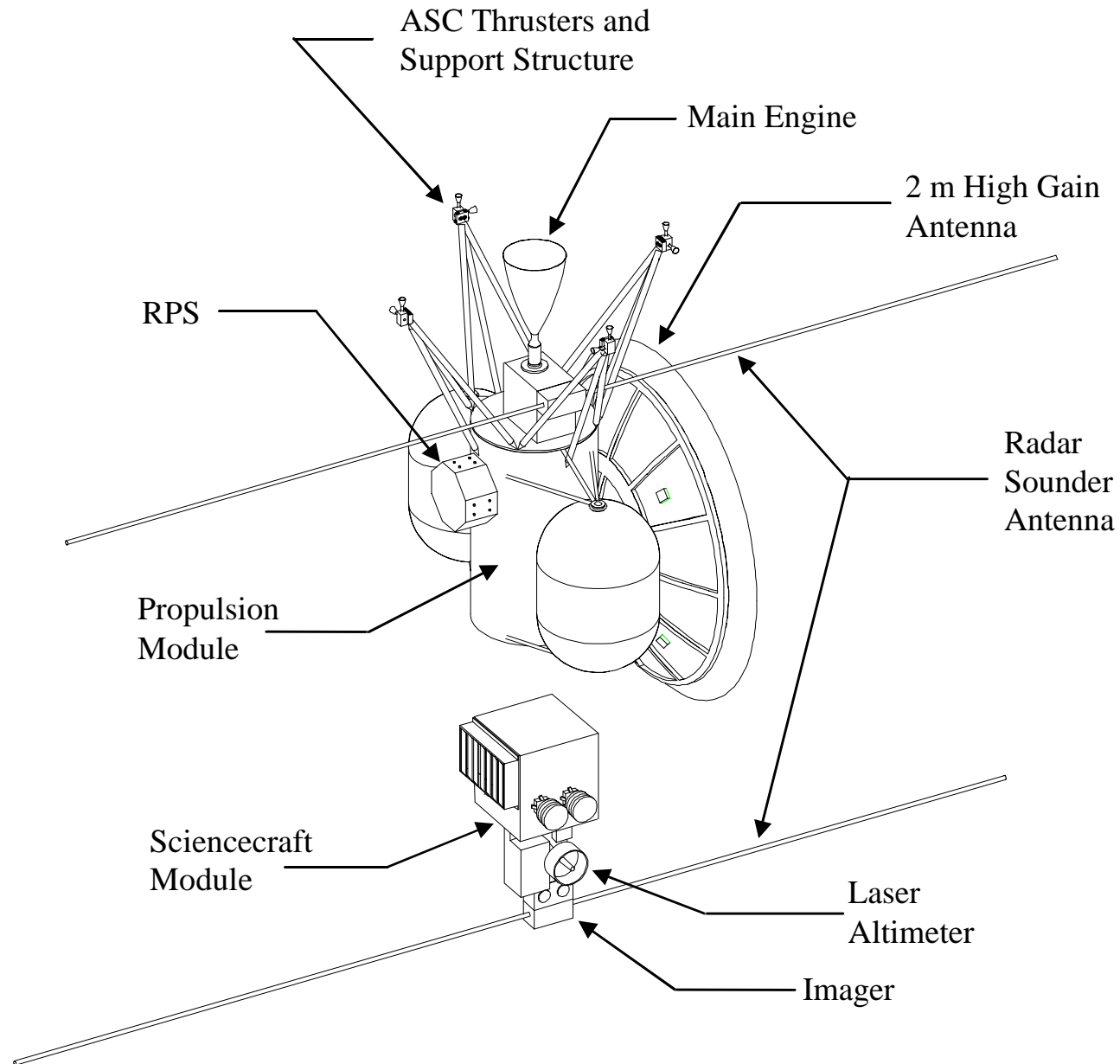
Pointing Control	2 mrad
Pointing Knowledge	1.5 mrad
Rate Control	<10 $\mu$ rad/sec
Processor Speed	4-50 MHz
Data Bus Rate	50 Mb/sec
Data Storage	Redundant 16 Gb
Downlink (Optical)	~100 Kb/sec @Europa
Power	150 W @Europa
$\Delta$ V Capability	2.5 km/sec

#### Legend:

MCM - Multichip Module  
MAC/NAC/WAC - Medium/Narrow/Wide Angle Camera  
DSTT - Tiny Deep Space Transponder  
ARPS - Advanced Radioisotope Power Source  
MLI - Multi-layer insulation  
RCS - Reaction Control System  
IMU - Inertial Measurement Unit  
RHU - Radioisotope Heater Unit

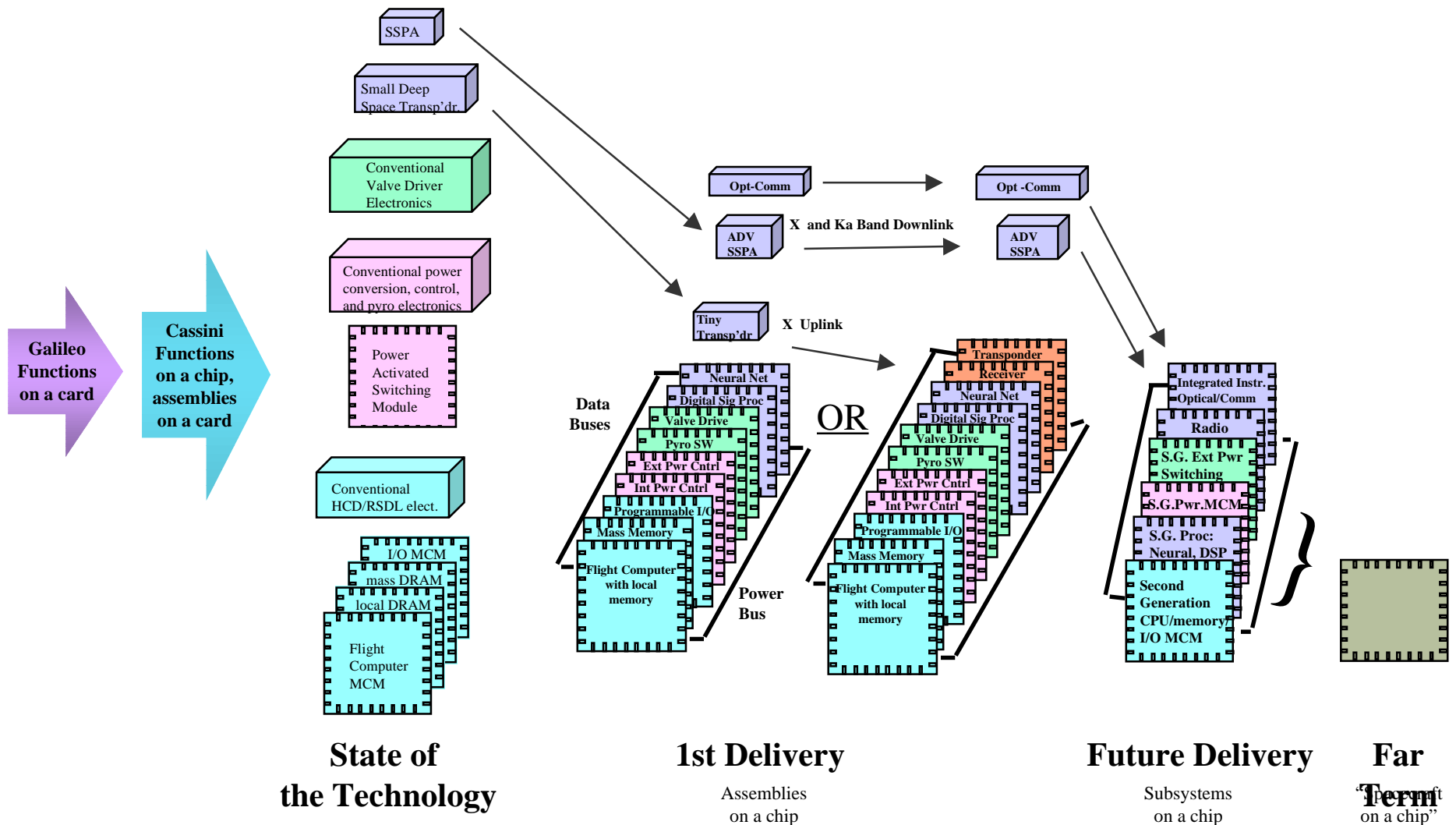
# Europa Orbiter Configuration using an RF telecom system

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# Advanced Deep Space System Development

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**HCD/RSDL** = Hardware Command Decoder/Need Solar Downlink

**MCM** = Multi-chip Modes

**NMP** = New Millennium Program

**OPT** = Optical

**DS1** = Deep Space 1, 1st NMP Technology Demo Flight

**PWR** = Power

**SSPA** = Solid State Power Amplifier

# Technology Development and Insertion

**X2000**



- Microelectronics
  - 3D MCM Stacking
  - Integrates C&DH, Attitude Control, Telecom, and Power & Pyro Switching
  - Multifunctional Structure
  - HDI power electronics
  - Provides a general purpose scaleable processing environment, inc:
    - Digital Signal Processors
    - Multi-processor architecture
    - Neural Nets
- Advanced Radioisotope Power Source
- Software
  - Unified flight and ground system architecture, employing flight and ground autonomy, tasks readily transfered between ground and flight
  - On-board, distributed applications and processing
  - On-board planning of flight system activities and navigation
  - Software Implemented Fault Tolerance
  - Scaleable to mission needs; a general platform for mission to build upon

**MCM** - Multichip Module    **HDI** - High Density Interconnects  
**C&DH** - Command & Data Handling

**COTS** - Commercial-off-the-shelf

# Technology Development and Insertion

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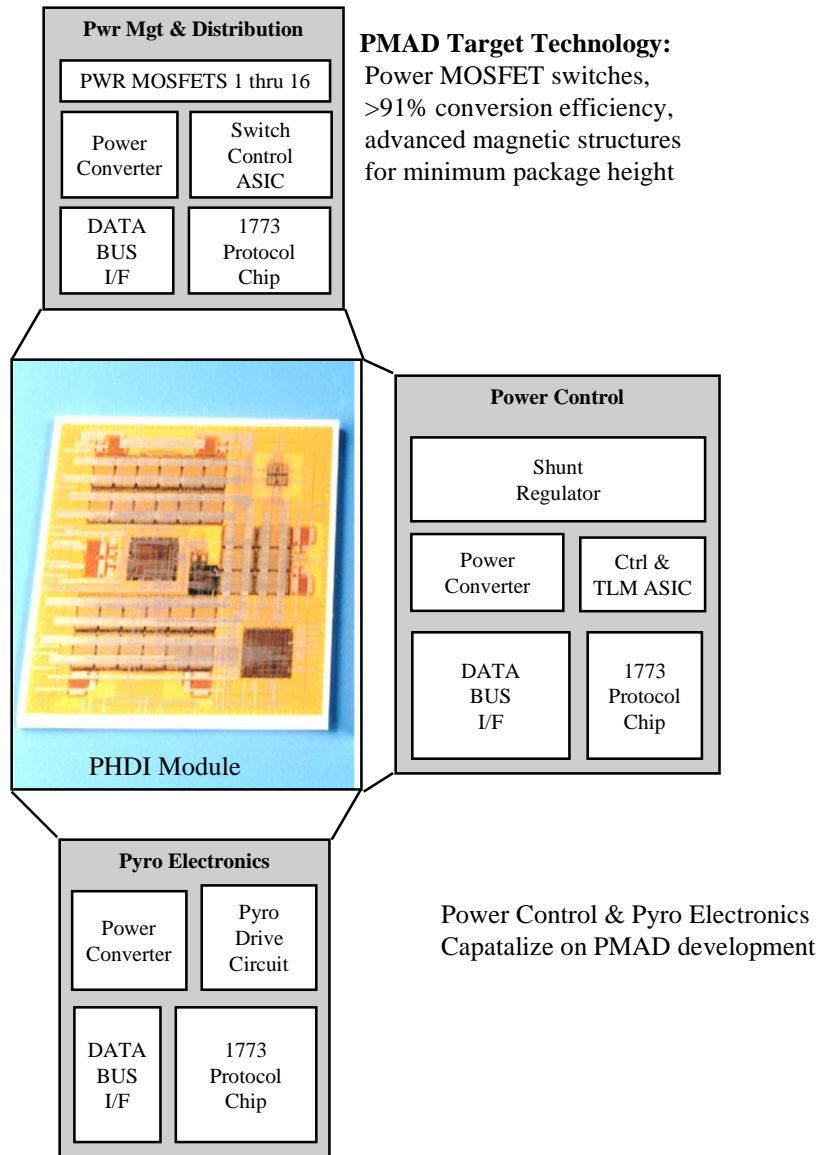
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- RF telecom
  - Transponder integrated with avionics
- Optical Communications
  - Comm Terminal capable of >100 kbps from Europa orbit
    - Also acts as laser altimeter receiver
    - Also acts as extremely high-resolution science imager
- Altitude control
  - Advanced Stellar Compass
- Propulsion
  - Hydrazine  $\mu$ thrusters
  - Variable liquid regulators
- Parts & Materials
  - Low voltage/power, rad hard electronic parts ( $\leq 3.3V$  for digital electronics)
  - Electronics parts list with radiation dose tolerances selectable by missions
  - Unshielded materials capable of withstanding radiation dose >25 Mrad
- Additional options
  - Next generation SEP
  - Optical processing



# Power Microelectronics Technologies



## Overview

**Current Technology:** PC Board packaging, discrete component control circuit, relay enable and command switches and SCR fire circuit for pyros

**Target Technologies:** 1. Mixed signal ASICs, 2. Power High Density Interconnect (PHDI) packaging, 3. modular fault tolerant design

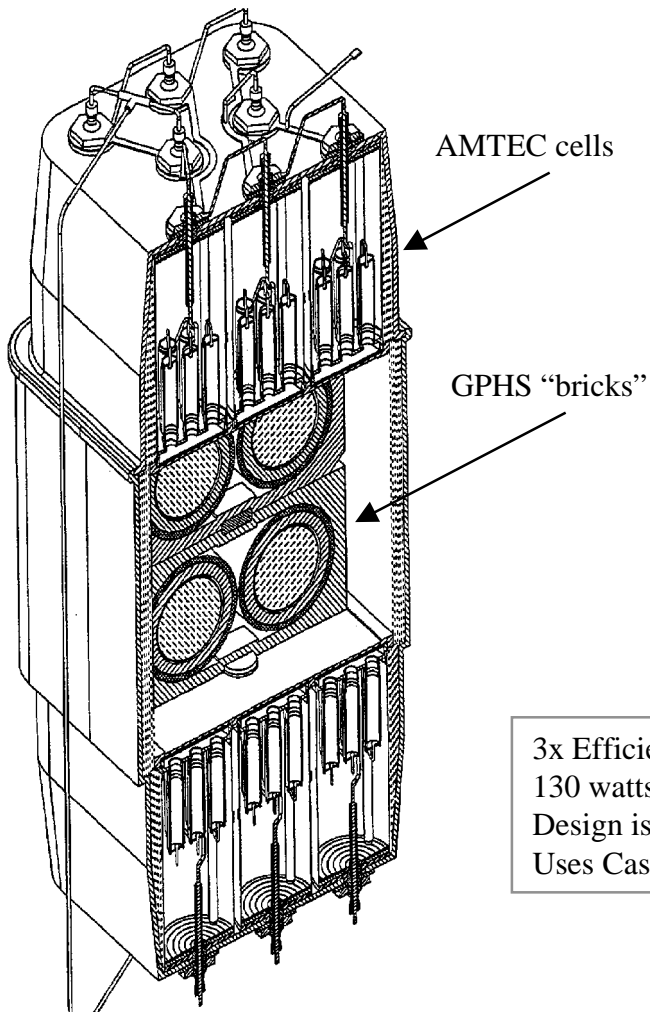
**Benefits:** 1. Reduce mass and volume, 2. standard command interface, 3. Provides peak power at maximum load, 4. configurable to various power source characteristics

**Impact of Fallback:** Several kg mass increase, increase in volume

**Key Issues:** Laser vs. NSI Pyro technology, thermal design, fault tolerant control circuit

# Advanced Radioisotope Power Source (ARPS) Technology

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## Overview

**Current Technology:** *Galileo/Cassini* heritage RTG with 6-7% efficiency  
unicouple conversion

**Target Technology:** >20% conversion using either AMTEC

**Benefits:** Mass reduction, smaller quantity of radioisotope

**Key Issues:** lifetime; efficiency; radiator size; heat rejection temperature

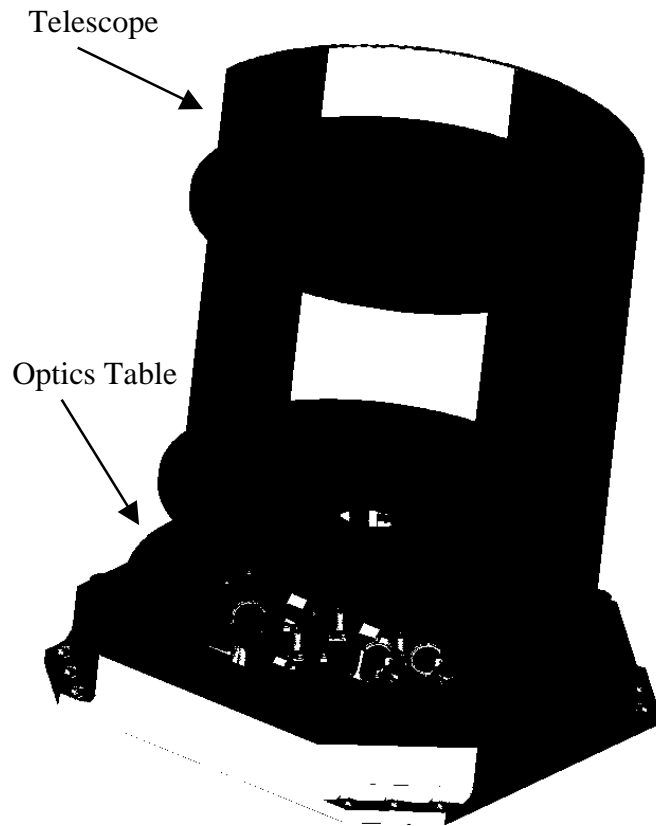
**Impact of Fallback:** > 12 kg mass increase

3x Efficiency over RTG's  
130 watts after 15 yrs  
Design is scalable in 50W Units  
Uses Cassini GPHS

**GPHS** = General Purpose Heat Source   **RTG** = Radioisotope Thermoelectric Generator

**AMTEC**= A thermally regenerative electro-chemical device converting heat to electricity directly using a solid electrolyte. It is sealed and has no moving parts. Advanced Modular Power Systems (AMPS) is a small business licensed to develop these devices commercially.

# Optical Communications Technology



## Overview

**Current Technology:** X/Ka-band RF telecommunications systems

**Target Technology:** 30-cm optical comm terminal, uplink & downlink

**Benefits:** Dramatic increase in telemetry rate at mass and power levels equivalent to RF telecom system; see plot below

**Key Issues:** lifetime, system-level fault protection

**Impact of Fallback:** Reduction in science data returned

Downlink, kbps  
(from Jovian Orbit)

